

Airworthiness Directive Schedule

Aeroplanes

Bolkow BO 208 C Junior

18 December 2025

- Notes:**
1. This AD schedule is applicable to Bolkow BO 208C aircraft manufactured under EASA Type Certificate No. A.358 (previously Luftfahrt-Bundesamt (LBA) Type Certificate No. 644).
 2. The European Union Aviation Safety Agency (EASA) is the National Airworthiness Authority (NAA) responsible for the issue of State of Design Airworthiness Directives (ADs) for these aircraft.

State of Design ADs can be obtained directly from the EASA website at:
<http://ad.easa.europa.eu/>
 3. The date above indicates the amendment date of this schedule.
 4. New or amended ADs are shown with an asterisk *

Contents

DCA/BOL/1	Nose Wheel Assembly - Modification	2	
DCA/BOL/3	Aileron Torque Tube - Replacement	2	
DCA/BOL/5	Elevator Spar - Modification	2	
DCA/BOL/6	Main Spar - Inspection	2	
DCA/BOL/7	Nose Gear Strut - Inspection.....	2	
DCA/BOL/8	Nose Gear Tube - Modification	2	
DCA/BOL/9	Fin Attachment Bolts - Inspection.....	2	
DCA/BOL/12	Fin Attachment - Modification.....	3	
DCA/BOL/13	Tachometer - Inspection	3	
DCA/BOL/14	Cancelled - Purpose fulfilled	3	
DCA/BOL/15	Engine Attachment Bolts - Modification.....	3	
DCA/BOL/16	Carburettor Hot Air Control - Modification	3	
DCA/BOL/17	Canopy Lock - Modification.....	3	
DCA/BOL/18	Engine Supports/Firewall - Inspection.....	3	
DCA/BOL/19	Nose Landing Gear - Inspection	3	
DCA/BOL/20	Wing, Aft Attachment - Inspection	4	
The State of Design ADs listed below are available directly from the National Airworthiness Authority (NAA) websites. Links to NAA websites are available on the CAA website at: Links to state of design airworthiness directives aviation.govt.nz If additional NZ ADs need to be issued when an unsafe condition is found to exist in an aircraft or aeronautical product in NZ, they will be added to the list below.			5
* 2025-0284	Rudder Control System / Drive - Inspection	5	

DCA/BOL/1 Nose Wheel Assembly - Modification

Applicability: Model 208 S/N 1 through 574
Requirement: Modify per Bolkow Junior SB 29A/64
Compliance: By 31 March 1965

DCA/BOL/3 Aileron Torque Tube - Replacement

Applicability: Model 208 S/N 505 through 566
Requirement: Replace torque tube per Bolkow Junior SB 33/65
Compliance: By 21 March 1966

DCA/BOL/5 Elevator Spar - Modification

Applicability: Model 208 S/N 509, 534, 537 and 566
Requirement: Modify per Bolkow Junior SB 34/65
Compliance: By 21 March 1966

DCA/BOL/6 Main Spar - Inspection

Applicability: Model 208C S/N 567, 571, 573, 574 and 576 through 601
Requirement: Inspect connection welds per Bolkow Junior SB 1/66
Compliance: At intervals not exceeding 100 hours TIS
Effective Date: 31 July 1966

DCA/BOL/7 Nose Gear Strut - Inspection

Applicability: All model 208 and 208C not modified per Bolkow Junior SB 9/65
Requirement: Inspect per Bolkow Junior SB 8/65
Compliance: At 500 hours TTIS and thereafter at intervals not exceeding 100 hours TIS until DCA/BOL/8 complied with
Effective Date: 31 July 1966

DCA/BOL/8 Nose Gear Tube - Modification

Applicability: All model 208 and 208C
Requirement: Modify per Bolkow Junior SB 9/65
Compliance: Within the next 500 hours TIS
Effective Date: 31 July 1966

DCA/BOL/9 Fin Attachment Bolts - Inspection

Applicability: All model 208 and 208C
Requirement: Inspect fin rear attachment bolts for correct type (AN4-6A) and torque. Check that AN960-416 washer fitted under nut and head of bolt
Compliance: At intervals not exceeding 100 hours TIS
Effective Date: 31 August 1968

DCA/BOL/12 Fin Attachment - Modification

Applicability: All model 208 and 208C
Requirement: Embody Bolkow modification AC33
Compliance: By 28 April 1968

DCA/BOL/13 Tachometer - Inspection

Applicability: All model 208 and 208C with VDO tachometer KI 12.200.11 or KI 12.200.21
Requirement: Check instrument for correct functioning and accuracy per VDO SL KI 12/L/1067.
Renew any instrument with an error in excess of allowable indication tolerance (+ 1.5% full-scale value).
Compliance: Within the next 10 hours TIS and thereafter at intervals not exceeding 3000 hours TIS
Effective Date: 31 August 1968

DCA/BOL/14 Cancelled - Purpose fulfilled**DCA/BOL/15 Engine Attachment Bolts - Modification**

Applicability: Model 208 and 208C S/N 505 through 684
Requirement: Modify per MBB SB 208-1/68
Compliance: By 2 May 1969

DCA/BOL/16 Carburettor Hot Air Control - Modification

Applicability: All model 208
Requirement: Embody Aerocraft (NZ) Ltd Mod. AC25A or approved equivalent
Compliance: By 31 August 1969

DCA/BOL/17 Canopy Lock - Modification

Applicability: All model 208
Requirement: Embody Aerocraft (NZ) Ltd Mod. AC51 or approved equivalent
Compliance: By 30 June 1969

DCA/BOL/18 Engine Supports/Firewall - Inspection

Applicability: All model 208
Requirement: Inspect per MBB SB 208-1/69
Compliance: At intervals not exceeding 100 hours TIS
Effective Date: 28 February 1970

DCA/BOL/19 Nose Landing Gear - Inspection

Applicability: All model 208 and 208C
Requirement: Using mirror and suitable light, visually inspect inside of nose gear outer tube for cracks after first accomplishing the following:
1. Models with hydraulic shock absorber (aircraft S/N 680 and lower)
(a) Loosen and remove lower shock absorber attachment.

(b) Secure aircraft in tail down attitude, remove nose wheel and ensure that springs are retained in correct position in inner tube.

2. Models with pneumatic shock absorber (aircraft S/N 681 and up)

(a) Remove lower torsion link spring attachment bolt and lower shock absorber bolt.

(b) Push down inner tube.

Replace any tube found cracked with new tube of same P/N per MBB SB 208-32/20-1 dated May 1972

Compliance: At intervals not exceeding 100 hours TIS

Effective Date: 30 September 1972

DCA/BOL/20 Wing, Aft Attachment - Inspection

Applicability: All model 208 series

Requirement: Accomplish visual inspection, dimensional check and dye penetrant inspection per MBB T1 208-1/80.

Rectify any defects found before further flight.

Compliance: Visual inspection and dimensional check - prior to next flight.

Dye penetrant inspection - within next 10 hours TIS

Effective Date: 28 March 1980

Note: Requirement notified to registered owners on effective date

NZCAR Part 111 leaflets B35-1, B35-1/1, B35-1/2, B35-1/3, B35-1/4 and B35-1/5 are hereby cancelled

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[Links to state of design airworthiness directives | aviation.govt.nz](#)

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*** 2025-0284 Rudder Control System / Drive - Inspection**

Applicability: Bölkow BO 208 C Junior and Bölkow Junior aircraft, all S/N.

Effective Date: 30 December 2025