Airworthiness Directive Schedule

Aeroplanes Extra EA 300 Series

Notes:

1 October 2020

- This AD schedule is applicable to Extra EA 300 series aircraft manufactured by Extra Flugzeugproduktions- und Vertriebs-GmdH under European Aviation Safety Agency (EASA) Type Certificate No. A.362.
- 2. The European Aviation Safety Agency (EASA) is the National Airworthiness Authority (NAA) responsible for the issue of State of Design Airworthiness Directives (ADs) for these aircraft. State of Design ADs can be obtained directly from the EASA website at http://ad.easa.europa.eu/

ADs issued by the Luftfahrt-Bundesamt (LBA) Germany, can be obtained directly from the LBA website at https://www2.lba.de/LTAs/

- 3. The date above indicates the amendment date of this schedule.
- 4. New or amended ADs are shown with an asterisk *

Contents

* LBA AD 1992-199 Elevator Mass Balance Support – Inspection	2
* LBA AD 1993-081 Rudder Control Cable Compression Sleeve – Inspection	2
* LBA AD 1993-145/2 Wing Root Rib/Auxiliary Spar Attchment - Inspection	2
* EASA AD D-1994-043R1 Upper Longeron Cutout Bridge – Inspection	2
* LBA AD 1994-258 Canopy Latches - Inspection	3
* LBA AD 1995-125 Air Inlet Screen - Inspection	3
* LBA AD 1995-126 Engine Cowling and Firewall Fire Protection – Inspection	3
* LBA AD 1995-443 Rudder Pedal Safety Control Stop – Inspection	3
* LBA AD 1997-203/2 Front Seat Belt Safety Cover – Inspection	4
* LBA AD 2002-408 Fuel Selector Valve – Inspection	4
* EASA AD D-2004-489 Bottom Fuselage Cover/Firewall Sealing – Inspection	4
* EASA AD 2006-0281 Upper Longeron at the Horizontal Stabiliser – Inspection	4
* EASA AD 2009-0160 Tail Spring Support – Inspection	5

Aeroplanes Extra EA 300 Series

* LBA AD 1992-199 Elevator Mass Balance Support - Inspection

Applicability: EA 300 aircraft, S/N V1 and 001 through to 034.

Note: Extra SB No. 300-1-92 dated 27 March 1992, or later approved revision pertains to

the subject of this AD.

Compliance: Initial compliance required before the issue of a New Zealand Certificate of

Airworthiness, or at the next Review of Airworthiness (RA), whichever is the sooner, unless previously accomplished. Repetitive inspections, if required, are to be accomplished at intervals not to exceed the times specified in the LBA AD.

Effective Date: 1 October 2020

* LBA AD 1993-081 Rudder Control Cable Compression Sleeve – Inspection

Applicability: EA 300 and EA 300/S aircraft, all S/N.

Note: Extra SB No. 300-1-93 dated 9 February 1993, or later approved revision pertains to

the subject of this AD.

Compliance: Initial compliance required before the issue of a New Zealand Certificate of

Airworthiness, or at the next Review of Airworthiness (RA), whichever is the sooner, unless previously accomplished. Repetitive inspections, if required, are to be accomplished at intervals not to exceed the times specified in the LBA AD.

Effective Date: 1 October 2020

* LBA AD 1993-145/2 Wing Root Rib/Auxiliary Spar Attachment - Inspection

Applicability: EA 300 aircraft, S/N V1 and 01 through to 45.

EA 300/S aircraft, S/N 01 through to 12.

Note: Extra SB No. 300-2-93 Issue C, or later approved revision pertains to the subject of

this AD.

Compliance: Initial compliance required before the issue of a New Zealand Certificate of

Airworthiness, or at the next Review of Airworthiness (RA), whichever is the sooner, unless previously accomplished. Repetitive inspections, if required, are to be accomplished at intervals not to exceed the times specified in the LBA AD.

Effective Date: 1 October 2020

* EASA AD D-1994-043R1 Upper Longeron Cutout Bridge – Inspection

Applicability: EA 300 aircraft, S/N V1 and 01 through to 50.

EA 300/S aircraft, S/N 01 through to 17.

Note: Extra SB No. 300-3-93 Issue B, or later approved revision pertains to the subject of

this AD.

Compliance: Initial compliance required before the issue of a New Zealand Certificate of

Airworthiness, or at the next Review of Airworthiness (RA), whichever is the sooner, unless previously accomplished. Repetitive inspections, if required, are to be accomplished at intervals not to exceed the times specified in the EASA AD.

Effective Date: 1 October 2020

* LBA AD 1994-258 Canopy Latches - Inspection

Applicability: EA 300/S aircraft, S/N 01 through to 24.

Note: Extra SB No. 300-3-94 Issue A, or later approved revision pertains to the subject of

this AD.

Compliance: Initial compliance required before the issue of a New Zealand Certificate of

Airworthiness, or at the next Review of Airworthiness (RA), whichever is the sooner, unless previously accomplished. Repetitive inspections, if required, are to be accomplished at intervals not to exceed the times specified in the LBA AD.

Effective Date: 1 October 2020

* LBA AD 1995-125 Air Inlet Screen - Inspection

Applicability: EA 300 aircraft, S/N V1 and 01 through to 51.

EA 300/S aircraft, S/N 01 through to 21.

Note: Extra SB No. 300-5-94 Issue A, or later approved revision pertains to the subject of

this AD.

Compliance: Initial compliance required before the issue of a New Zealand Certificate of

Airworthiness, or at the next Review of Airworthiness (RA), whichever is the sooner, unless previously accomplished. Repetitive inspections, if required, are to be accomplished at intervals not to exceed the times specified in the LBA AD.

Effective Date: 1 October 2020

* LBA AD 1995-126 Engine Cowling and Firewall Fire Protection - Inspection

Applicability: EA 300 aircraft, S/N V1 and 01 through to 53.

EA 300/S aircraft, S/N 01 through to 22.

Note: Extra SB No. 300-6-94 Issue A, or later approved revision pertains to the subject of

this AD.

Compliance: Initial compliance required before the issue of a New Zealand Certificate of

Airworthiness, or at the next Review of Airworthiness (RA), whichever is the sooner, unless previously accomplished. Repetitive inspections, if required, are to be accomplished at intervals not to exceed the times specified in the LBA AD.

Effective Date: 1 October 2020

* LBA AD 1995-443 Rudder Pedal Safety Control Stop - Inspection

Applicability: EA 300 aircraft fitted with an electric actuated rudder pedal adjustment produced up

to November 1995. An electric actuated rudder pedal adjustment modification could

have been installed at manufacture, or embodied in service.

EA 300/S aircraft, S/N 001 through to 028. EA 300/L aircraft, S/N 001 through to 015.

Note: Extra SB No. 300-3-95 Issue A, or later approved revision pertains to the subject of

this AD.

1 October 2020

Compliance: Initial compliance required before the issue of a New Zealand Certificate of

Airworthiness, or at the next Review of Airworthiness (RA), whichever is the sooner, unless previously accomplished. Repetitive inspections, if required, are to be

accomplished at intervals not to exceed the times specified in the LBA AD.

Effective Date:

* LBA AD 1997-203/2 Front Seat Belt Safety Cover – Inspection

Applicability: EA 300/200 aircraft, S/N 01 through to 15.

Note: Extra SB No. 300-1-97 Issue B, or later approved revision pertains to the subject of

this AD.

Compliance: Initial compliance required before the issue of a New Zealand Certificate of

Airworthiness, or at the next Review of Airworthiness (RA), whichever is the sooner, unless previously accomplished. Repetitive inspections, if required, are to be accomplished at intervals not to exceed the times specified in the LBA AD.

Effective Date: 1 October 2020

* LBA AD 2002-408 Fuel Selector Valve - Inspection

Applicability: EA 300/S, EA 300/L and EA 200 aircraft, all S/N.

Note: Extra SL No. 300-09-02 Issue A, or later approved revision pertains to the subject of

this AD.

Compliance: Initial compliance required before the issue of a New Zealand Certificate of

Airworthiness, or at the next Review of Airworthiness (RA), whichever is the sooner, unless previously accomplished. Repetitive inspections, if required, are to be accomplished at intervals not to exceed the times specified in the LBA AD.

Effective Date: 1 October 2020

* EASA AD D-2004-489 Bottom Fuselage Cover/Firewall Sealing - Inspection

Applicability: Part 1 of AD D-2004-489:

EA 300 aircraft, S/N 0 through to 67. EA 300/S aircraft, S/N 0 through to 31.

EA 330/L aircraft, S/N 0 through to 167, 168 through to 170 (or converted to 1168 through to 1170), 1171, 172 (or converted to 1172), 173 (or converted to 1173) and

1174 through to 1181.

EA 300/200 aircraft, S/N 0 through to 31.

Part II of AD D-2004-489:

EA 300 aircraft, EA 300/S aircraft, EA 300/L aircraft and EA 300/200 aircraft, all S/N.

Note: Extra SB No. 300-4-04 Issue A, or later approved revision pertains to the subject of

this AD.

Compliance: Initial compliance required before the issue of a New Zealand Certificate of

Airworthiness, or at the next Review of Airworthiness (RA), whichever is the sooner, unless previously accomplished. Repetitive inspections, if required, are to be accomplished at intervals not to exceed the times specified in the EASA AD.

Effective Date: 1 October 2020

* EASA AD 2006-0281 Upper Longeron at the Horizontal Stabiliser – Inspection

Applicability: EA 300 series aircraft, all S/N.

Note: Extra SB No. 300-2-95 Issue F, or later approved revision pertains to the subject of

this AD.

Compliance: Initial compliance required before the issue of a New Zealand Certificate of

Airworthiness, or at the next Review of Airworthiness (RA), whichever is the sooner, unless previously accomplished. Repetitive inspections, if required, are to be accomplished at intervals not to exceed the times specified in the EASA AD.

Effective Date: 1 October 2020

* EASA AD 2009-0160 Tail Spring Support - Inspection

Applicability: EA 300/L aircraft, S/N 01 through to 170, 172, 173, 1171 and 1174 through to 1299.

EA 300/200 aircraft, S/N 01 through to 31 and 1032 through to 1043.

Note: Extra SB No. 300-2-97 Issue B, or later approved revision pertains to the subject of

this AD.

Compliance: Initial compliance required before the issue of a New Zealand Certificate of

Airworthiness, or at the next Review of Airworthiness (RA), whichever is the sooner,

unless previously accomplished. Repetitive inspections, if required, are to be accomplished at intervals not to exceed the times specified in the EASA AD.

Effective Date: 1 October 2020