Airworthiness Directive Schedule

Aeroplanes Mitsubishi MU-2B-30 Series 25 June 2020

Notes

- 1. This AD schedule is applicable to Mitsubishi Heavy Industries Ltd. MU-2B-30 aircraft manufactured under Japan Civil Aviation Bureau (JCAB) Type Certificate No. 25.
- 2. The Japan Civil Aviation Bureau (JCAB) is the National Airworthiness Authority (NAA) responsible for the issue of State of Design Airworthiness Directives (ADs) for these aircraft. JCAB ADs can be obtained directly from the JCAB website at http://www.mlit.go.jp/koku/english/index.html
- 3. The date above indicates the amendment date of this schedule.
- 4. New or amended ADs are shown with an asterisk *

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* DCA/MU2B-30/1A Airworthiness Directive Compliance

Applicability: Model MU-2B-30 aircraft, all S/N.

Requirement: Compliance with the following Japan Civil Aviation Bureau (JCAB) Airworthiness

Directives (as applicable) is required:

TCD-518-68 Elevator Hinge
TCD-642-70 Fuel Anti-Ice Additive
TCD-643-70 Fuel Quantity Indicator
TCD-653-70 Nacelle Structure
TCD-859-72 Battery System
TCD-866-72 NLG Actuator
TCD-947A-74 Flap Flexible Shafts

TCD-1188-75 Propeller Pitch Control Lever TCD-1202-75 Engine Nacelle Door Latches

TCD-1209-75 NLG Shimmy Damper TCD-1094A-76 Flap Jack Screw MLG Drag Strut

TCD-1328-76 Rear Baggage Compartment

TCD-1370-76 Wing Tip Tank Fuel

TCD-1520A-77 Engine Nacelle Door Latch Link TCD-2048-81 Engine Fire Detection System

TCD-2070-81 NLG Retraction
TCD-2263-83 NLG Drag Strut Bracket
TCD-1001B-2-84 Elevator Trim Tab Bracket
TCD-2342-1-84 Engine Air Inlet Anti-ice System
TCD-2450A-87 Pitot Tubes and Anti-ice Circuits

TCD-2450A-87 Pitot Tubes and An TCD-2451-1 Flap Flexible Shaft Autopilot System

TCD-2545-1-88 Generator Ground Circuit

TCD-3378-90 Rudder Trim Tab
TCD-3464-91 Autopilot System
TCD-3901-93 Wing Tip Tank Fuel

TCD-3942-1-94 Emergency Geardown Gearbox

TCD-3393A-96 Anti-ice/Deice Systems

TCD-4310-96 Wing Tip Tank

TCD-4552-96 Severe Icing Conditions TCD-2679A-97 Engine Auto-ignition System

TCD-2856-1-97 Autopilot System
TCD-4529-97 Ice Detection System
TCD-4626-97 De-ice Monitoring System

TCD-4684-97 Landing Gear Actuating Mechanism TCD-3740A-98 Autopilot System and Nose down Trim

TCD-4838-98 Landing Gear Drive Train

TCD-4889-98 Engine Torque Meter Calibration

TCD-4890-98 Flight Idle Fuel Flow
TCD-5432-2000 Pneumatic Deicing Boots
TCD-4625A-2001 Trim-in-motion Alert System

TCD-5547-2001 Power Levers

TCD-5747-2001 Tip Tank Attach Brackets

TCD-5864-2001 Flap Drive Train TCD-6127-2003 Pneumatic Lines

* TCD-4753A-2004 Cancelled - TCD-4753B-2020 refers

TCD-7030-2006 Outer Wing

TCD-4889-2-2007 Engine Torque Meter Calibration

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Aeroplanes Mitsubishi MU-2B-30 Series

Note: Each part of this AD (each individual JCAB AD) shall be certified in the aircraft log

book separately.

Compliance: Initial compliance required before the issue of a New Zealand Certificate of

Airworthiness, or at the next Review of Airworthiness (RA), whichever is the sooner, unless previously accomplished. Repetitive inspections, if required, are to be accomplished at intervals not to exceed the times specified in the JCAB AD.

Effective Date: DCA/MU2B-30/1 - 27 August 2009

DCA/MU2B-30/1A - 10 June 2020

DCA/MU2B-30/2 MLG Forward Door Emergency Cable – Inspection

Applicability: Model MU-2B-30 aircraft, S/N 502 through to 547 not fitted with MLG forward door

emergency cable P/N 030A-38558-41/-51.

Requirement: To prevent failure of the MLG forward door emergency operating cable, accomplish

the requirements specified in JCAB AD TCD-1235-75.

Note 1: Mitsubishi MU-2 SB No. 170 or later JCAB approved revisions pertains to the subject

of this AD.

Note 2: A copy of the JCAB AD is available on the JCAB web site at

http://www.mlit.go.ip/koku/15 hf 000127.html

Note 3: Replacement of MLG forward door emergency cable with P/N 030A-38558-41/-51 per

the instructions in SB No. 170 is a terminating action to the repetitive inspection

requirements of this AD.

(JCAB AD TCD-1235-75 refers)

Compliance: At the initial threshold specified in JCAB AD TCD-1235-75 unless previously

accomplished, and thereafter at intervals not to exceed those specified in the JCAB

AD.

Effective Date: 27 August 2009

DCA/MU2B-30/3 MLG Forward Door Mechanism – Inspection

Applicability: Model MU-2B-30 aircraft, S/N 501 through to 585.

Requirement: To prevent failure of support P/N 030A-38824 of the MLG forward door actuating

mechanism, accomplish the requirements specified in JCAB AD TCD-1272-75.

Note 1: Mitsubishi MU-2 SB No. 172 or later JCAB approved revisions pertains to the subject

of this AD.

Note 2: A copy of the JCAB AD is available on the JCAB web site at

http://www.mlit.go.jp/koku/15_hf_000127.html

(JCAB AD TCD-1272-75 refers)

Compliance: At the initial threshold specified in JCAB AD TCD-1272-75 unless previously

accomplished, and thereafter at intervals not to exceed those specified in the JCAB

AD.

Effective Date: 27 August 2009

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DCA/MU2B-30/4 Front Windscreen – Inspection

Applicability: Model MU-2B-30 aircraft, S/N 501 through to 673 except 313, 321, 651 and 652.

Note 1: This AD is not applicable to aircraft already in compliance with Mitsubishi MU-2

Service Recommendation No. 027 or SB No. 163B.

Requirement: To prevent failure of front windshield due to an inoperative air conditioning system,

accomplish the requirements specified in JCAB AD TCD-948A-1-78.

Note 2: Mitsubishi MU-2 SB No. 163B and Service Recommendation No. 027A or later JCAB

approved revisions pertains to the subject of this AD.

Note 3: A copy of the JCAB AD is available on the JCAB web site at

http://www.mlit.go.jp/koku/15_hf_000127.html

Note 4: The installation of a defogging air temp warning system per SB No. 163B is a

termination action to the repetitive inspection requirements of this AD.

Note 5: This AD partially revises JCAB AD 948A-74.

(JCAB AD TCD-948A-1-78 refers)

Compliance: At the initial threshold specified in JCAB AD TCD-948A-1-78 unless previously

accomplished, and thereafter at intervals not to exceed those specified in the JCAB

AD.

Effective Date: 27 August 2009

DCA/MU2B-30/5 Cancelled – DCA/MU2B-30/9 refers

Effective Date: 28 June 2012

DCA/MU2B-30/6 Elevator Trim Tab Bracket – Inspection

Applicability: Model MU-2B-30 aircraft, S/N 501 through to 696 except 652 and 661.

Requirement: To prevent failure of the elevator trim tab bracket due to possible wear, accomplish

the requirements specified in JCAB AD TCD-1001B-81.

Note 1: Mitsubishi MU-2 SB No. 176C or later JCAB approved revisions pertains to the

subject of this AD.

Note 2: A copy of the JCAB AD is available on the JCAB web site at

http://www.mlit.go.jp/koku/15_hf_000127.html

Note 3: The installation of bracket P/N 035A-22128-1/-2 is a terminating action to the

repetitive inspection and lubrication requirements of this AD.

Note 4: This AD supersedes JCAB TCD-1001A-76 and TCD-1001A-1-80.

(JCAB AD TCD-1001B-81 refers)

Compliance: At the initial threshold specified in JCAB AD TCD-1001B-81 unless previously

accomplished, and thereafter at intervals not to exceed those specified in the JCAB

AD.

Effective Date: 27 August 2009

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DCA/MU2B-30/7 Tip Tank Fuel – Inspection

Applicability: Model MU-2B-30 aircraft, S/N 501, 504 and 548 through to 673 except 652 and 661.

Requirement: To prevent fuel from the tip tank from entering into the strobe light, accomplish the

requirements specified in JCAB AD TCD-1370A-81.

Note 1: Mitsubishi MU-2 SB No. 174C or later JCAB approved revisions pertains to the

subject of this AD.

Note 2: A copy of the JCAB AD is available on the JCAB web site at

http://www.mlit.go.jp/koku/15_hf_000127.html

Note 3: The embodiment of modification per instruction 2 of SB No. 174C is a terminating

action to the repetitive inspection requirements of this AD.

Note 4: This AD supersedes JCAB TCD-1370-76.

(JCAB AD TCD-1370A-81 refers)

Compliance: At the initial threshold specified in JCAB AD TCD-1370A-81 unless previously

accomplished, and thereafter at intervals not to exceed those specified in the JCAB

AD.

Effective Date: 27 August 2009

DCA/MU2B-30/8 Engine Electrical Harness – Inspection

Applicability: Model MU-2B-30 aircraft, S/N 501 through to 696 excluding 652 and 661.

Requirement: To prevent failure of wires P/N MS 25471 in the engine electrical harness, accomplish

the requirements specified in JCAB AD TCD-2113-82.

Note 1: Mitsubishi MU-2 SB No. 191 or later JCAB approved revisions pertains to the subject

of this AD.

Note 2: A copy of the JCAB AD is available on the JCAB web site at

http://www.mlit.go.jp/koku/15_hf_000127.html

Note 3: The replacement of wires P/N MS 25471 in the electrical harness with P/N M22759

per paragraph 3 of SB No. 191 is a terminating action to the repetitive inspection

requirements of this AD.

(JCAB AD TCD-2113-82 refers)

Compliance: At the initial threshold specified in JCAB AD TCD-2113-82 unless previously

accomplished, and thereafter at intervals not to exceed those specified in the JCAB

AD.

DCA/MU2B-30/9 NLG Strut Assembly – Inspection

Applicability: Model MU-2B-30 aircraft, S/N 501 through to 696 excluding 652 and 661.

Requirement: To prevent failure of the NLG strut assembly, accomplish the requirements specified

in JCAB AD TCD-1768-1-85.

Note 1: Mitsubishi MU-2 SB No. 181 revision B or later JCAB approved revisions pertains to

the subject of this AD.

Note 2: A copy of the JCAB AD is available on the JCAB web site at

http://www.mlit.go.jp/koku/15_hf_000127.html

Note 3: The replacement of the NLG strut assembly with parts marked with 'SP' per part II of

SB No. 181 is a terminating action to the repetitive inspection requirements of this AD.

(JCAB AD TCD-1768-1-85 refers)

Compliance: At the initial threshold specified in JCAB AD TCD-1768-1-85 unless previously

accomplished, and thereafter at intervals not to exceed those specified in the JCAB

AD.

Effective Date: 27 August 2009

DCA/MU2B-30/10 Fuel Tanks – Inspection

Applicability: Model MU-2B-30 aircraft, S/N 502 through to 551.

Requirement: To prevent blocked fuel lines due to possible peeling of the DV1180 fungus resistant

paint in the wing fuel tanks, accomplish the requirements specified in JCAB AD TCD-

730-2-86.

Note 1: Mitsubishi MU-2 SB No. 143C or later JCAB approved revisions pertains to the

subject of this AD.

Note 2: A copy of the JCAB AD is available on the JCAB web site at

http://www.mlit.go.jp/koku/15 hf 000127.html

Note 3: A complete recoating of the fuel tanks per the instructions in SB No. 143C is a

terminating action to the repetitive inspection requirements of this AD.

(JCAB AD TCD-730-2-86 refers)

Compliance: At the initial threshold specified in JCAB AD TCD-730-2-86 unless previously

accomplished, and thereafter at intervals not to exceed those specified in the JCAB

AD.

DCA/MU2B-30/11 Flap Control System – Inspection

Applicability: Model MU-2B-30 aircraft, S/N 501 through to 696 excluding 652 and 661.

Requirement: To prevent failure of the flap torque tube assembly joint, accomplish the requirements

specified in JCAB AD TCD-2252-2-88.

Note 1: Mitsubishi MU-2 SB No. 189B or later JCAB approved revisions pertains to the subject

of this AD.

Note 2: A copy of the JCAB AD is available on the JCAB web site at

http://www.mlit.go.jp/koku/15_hf_000127.html

Note 3: This AD supersedes JCAB AD TCD-2252-1-87.

(JCAB AD TCD-2252-2-88 refers)

Compliance: At the initial threshold specified in JCAB AD TCD-2252-2-88 unless previously

accomplished, and thereafter at intervals not to exceed those specified in the JCAB

AD.

Effective Date: 27 August 2009

DCA/MU2B-30/12 Cockpit and Cabin Windows – Inspection

Applicability: Model MU-2B-30 aircraft, S/N 501 through to 696 excluding 652 and 661.

Requirement: To prevent failure of the cockpit windscreen, cockpit side windows and cabin windows,

accomplish the requirements specified in JCAB AD TCD-4311-95.

Note 1: Mitsubishi MU-2 SB No. 224A or later JCAB approved revisions pertains to the subject

of this AD.

Note 2: A copy of the JCAB AD is available on the JCAB web site at

http://www.mlit.go.jp/koku/15_hf_000127.html

(JCAB AD TCD-4311-95 refers)

Compliance: At the initial threshold specified in JCAB AD TCD-4311-95 unless previously

accomplished, and thereafter at intervals not to exceed those specified in the JCAB

AD.

DCA/MU2B-30/13 Propeller Feathering – Inspection

Applicability: Model MU-2B-30 aircraft, S/N 501 through to 696 excluding 652 and 661.

Requirement: To prevent the inability to feather the propeller due to possible misadjustment of the

feathering linkage, accomplish the requirements specified in JCAB AD TCD-4379-96.

Note 1: Mitsubishi MU-2 SB No. 229 or later JCAB approved revisions pertains to the subject

of this AD.

Note 2: A copy of the JCAB AD is available on the JCAB web site at

http://www.mlit.go.jp/koku/15_hf_000127.html

(JCAB AD TCD-4379-96 refers)

Compliance: At the initial threshold specified in JCAB AD TCD-4379-96 unless previously

accomplished, and thereafter at intervals not to exceed those specified in the JCAB

AD.

Effective Date: 27 August 2009

DCA/MU2B-30/14 Fuselage – Inspection

Applicability: Model MU-2B-30 aircraft, S/N 501 through to 696 excluding 652 and 661.

Requirement: To prevent the inability to feather the propeller due to possible misadjustment of the

feathering linkage, accomplish the requirements specified in JCAB AD TCD-5030-99.

Note 1: Mitsubishi MU-2 SB No. 209B or later JCAB approved revisions pertains to the subject

of this AD.

Note 2: A copy of the JCAB AD is available on the JCAB web site at

http://www.mlit.go.jp/koku/15_hf_000127.html

(JCAB AD TCD-5030-99 refers)

Compliance: At the initial threshold specified in JCAB AD TCD-5030-99 unless previously

accomplished, and thereafter at intervals not to exceed those specified in the JCAB

AD.

The State of Design ADs listed below are available directly from the National Airworthiness Authority (NAA) websites. Links to NAA websites are available on the CAA website at http://www.caa.govt.nz/airworthiness-directives/states-of-design/

If additional NZ ADs need to be issued when an unsafe condition is found to exist in an aircraft or aeronautical product in NZ, they will be added to the list below.

TCD-8231-2013 Airframe – Inspection

Effective Date: 20 August 2013

TCD-8595-2015 MLG Oleo Attach Fittings – Inspection

Effective Date: 15 July 2015

TCD-8678-2016 Propeller Pitch Control Lever – Inspection

Effective Date: 19 February 2016

TCD-8783-2016Wing Spacer Plates - Inspection

Effective Date: 12 July 2016

* TCD-4753B-2020 Pilot in Charge Training Requirements

Applicability: Mitsubishi MU-2B and MU-2B-30 series aircraft, all S/N.

Effective Date: 10 June 2020