In addition to the ADs scheduled below, relevant ADs in the ‘AIRCRAFT GENERAL’ AD Schedule and any other relevant ADs in Section C ‘COMPONENTS AND EQUIPMENT’ shall be complied with.

**DCA/DH104/1**  
**Mandatory Modifications**  
**Applicability:** All model 104  
**Requirement:** Embody all applicable UK CAA mandatory modifications  
**Compliance:** Prior to initial C of A issue

**DCA/DH104/2**  
**Centre Section and Wing - Safe Life**  
**Applicability:** All model 104  
**Requirement:** Replace centre section and wing lower spar boom components per Hawker Siddeley TNS 119  
**Compliance:** As detailed in TNS 119  
**Effective Date:** 19 July 1971

**DCA/DH104/3**  
**Wing Assembly - Inspection**  
**Applicability:** All model 104 in which mod 852 was not embodied during construction  
**Requirement:** Inspect wing spar booms for corrosion per Hawker Siddeley TNS 125  
**Compliance:** As detailed in TNS 125  
**Effective Date:** 26 June 1967

**DCA/DH104/4**  
**Wing Spar Booms - Inspection**  
**Applicability:** Model 104 up to S/N 04465 without mod. 870 embodied  
**Requirement:** X-ray per Hawker Siddeley TNS 134  
**Compliance:** As detailed in TNS 134  
**Effective Date:** 17 March 1975
DCA/DH104/5  Wing Lower Root Joint - Inspection  
Applicability: All model 104  
Requirement: Inspect fittings per Hawker Siddeley TNS 168  
Compliance: At intervals not exceeding 4500 hours TIS or 6 years, whichever is the sooner  
Effective Date: 12 July 1971

DCA/DH104/6  Control Column - Inspection  
Applicability: All model 104  
Requirement: Inspect per Hawker Siddeley TNS 171  
Compliance: Whenever aileron control circuit in column is disturbed  
Effective Date: 11 July 1960

DCA/DH104/7  Trim Tab Controls - Inspection  
Applicability: All model 104  
Requirement: Inspect rudder and elevator connecting rod eye ends P/N CM 2A and CM 28 per Hawker Siddeley TNS 187  
Compliance: At intervals not exceeding 2400 hours TIS  
Effective Date: 24 February 1964

DCA/DH104/8  Engine Mounts - Inspection  
Applicability: All model 104 with engine mounts P/N 4 EM 201A and 203A  
Requirement: X-ray per Hawker Siddeley TNS 190  
Compliance: At intervals not exceeding 4 years  
Effective Date: 12 January 1970

DCA/DH104/9  Rudder Control System - Inspection  
Applicability: All model 104  
Requirement: Inspect reversal lever P/N 4CF 767A per Hawker Siddeley TNS 219  
Compliance: At intervals not exceeding 2400 hours TIS  
Effective Date: 19 October 1970

DCA/DH104/10  Engine Mount Attachment - Inspection  
Applicability: All model 104  
Requirement: Inspect attachment fittings per Hawker Siddeley TNS 226  
Compliance: As detailed in TNS 226  
Effective Date: 25 October 1971
DCA/DH104/11  Engine Mount - Inspection
Applicability:  All model 104
Requirement:  Inspect LH mounting frame P/N 4EM 201A per Hawker Siddeley TNS 228
Compliance:  At intervals not exceeding 75 hours TIS
Effective Date:  17 April 1972

DCA/DH104/12  Nose Undercarriage - Inspection
Applicability:  All model 104
Requirement:  Inspect inner casing per Hawker Siddeley TNS 232
Compliance:  At intervals not exceeding 600 hours TIS until mod. 1728 embodied and thereafter at intervals not exceeding 2400 hours TIS
Effective Date:  17 September 1973

DCA/DH104/13  Centre Section - Inspection
Applicability:  All model 104
Requirement:  Inspect centre section main spar top boom P/N 4FS. 135 A/1 per Hawker Siddeley TNS 237
Compliance:  As detailed in TNS 237
Effective Date:  9 September 1976

DCA/DH104/14  Centre Section - Inspection
Applicability:  All model 104 without mod. 779 embodied
Requirement:  Inspect main spar bottom boom per Hawker Siddeley TNS 238
Compliance:  As detailed in TNS 238
Effective Date:  24 January 1977

DCA/DH104/15  Aircraft Life Extension - Modification
Applicability:  All model 104
Requirement:  Modify per British Aerospace TNS 240
Compliance:  When mod. 1707 is embodied for extension of aircraft life

DCA/DH104/16  Engine Mount - Inspection
Applicability:  All model 104
Requirement:  Inspect front engine support spigot P/N 4EM 681 per British Aerospace TNS 244
Compliance:  By 30 November 1978
DCA/DH104/17  Aileron Balance Weight Assembly - Inspection

Applicability: All model 104

Requirement: Accomplish visual and x-ray inspections per British Aerospace TNS 245 Issue 2.

Assemblies with internal or external corrosion exceeding stated limits must be removed from service before further flight.

Note: Radiographs are to be sent to British Aerospace for interpretation as prescribed in TNS 245 Issue 2 and results notified to CAD when received

Compliance: Visual inspection - within next 14 days
X-ray inspection - within next 50 hours TIS or 28 days, whichever is the sooner

Effective Date: 25 June 1987

DCA/DH104/18  Flap Jack to Flap Attachment - Inspection

Applicability: All model 104

Requirement: To prevent possible asymmetric flap operation due to installation of an unapproved, non-safety locked pin, visually inspect each flap installation to ensure that correct pin P/N 4 WF 329 is fitted and correctly locked.

British Aerospace TNS 246 refers

Compliance: By 26 December 1988

Effective Date: 19 December 1988

Note: 1. Requirement notified to registered owners on effective date.
2. Copy of reference document may be obtained from the Director.