Airworthiness Directive Schedule
Aeroplanes
Rallye Series
22 October 2015

Notes

1. This AD schedule is applicable to the following aircraft models manufactured by Société de Construction d’Avions de Tourisme et d’Affaires "S.O.C.A.T.A." - Groupe AEROSPATIALE, Société d'Exploitation des Etablissements MORANE-SAULNIER, Société MORANE-SAULNIER under Type Certificate Numbers:

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2. The European Aviation Safety Agency (EASA) is the National Airworthiness Authority (NAA) responsible for the issue of State of Design Airworthiness Directives (ADs) for these aircraft. State of Design ADs can be obtained directly from the EASA web site. The link to the EASA web site is available on the CAA web site at http://www.caa.govt.nz/Airworthiness_Directives/states_of_design.html

3. The date above indicates the amendment date of this schedule.

4. New or amended ADs are shown with an asterisk *

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DCA/MS/110  Cancelled: Once only inspection, purpose fulfilled

DCA/MS/111  Horizontal Stabiliser Fittings - Inspection
Applicability: All model 880B and 885
Requirement: Socata SB 40/1
Compliance: Every periodic inspection
Effective Date: 30 September 1964

DCA/MS/112  Leading Edge Slat - Inspection
Applicability: All model 880B and 885
Requirement: Morane Saulnier SB 14
Compliance: Every alternate periodic inspection
Effective Date: 30 September 1964

DCA/MS/113  Aileron Control Tube - Inspection
Applicability: Model 880B S/N 1 through 264
Requirement: Morane Saulnier SB 19
Compliance: As detailed
Effective Date: 30 September 1964

DCA/MS/114A  Landing Gear - Inspection
Applicability: All model 880B and 885
Requirement: Morane Saulnier SB 42
Compliance: 1. Every periodic inspection and after a heavy landing
2. Dye penetrant check to be carried out on main landing gear every 100 hours TIS
Effective Date: 30 September 1966

DCA/MS/115  Rudder Pedal Cracking - Inspection
Applicability: All model 880B and 885
Requirement: Cracks have been found in the region of the welded joint of the rudder pedal assembly hinge lug to the torque tube P/N 880-27-0-093-2.
Visually inspect the welded joint in the region of the rudder pedal assembly hinge lug to the torque tube for cracks.
Compliance: At the next periodic inspection and at each subsequent periodic inspection
Effective Date: 30 September 1966

DCA/MS/116  Aileron Control - Inspection
Applicability: All model 880B and 885
Requirement: Morane Saulnier SB 33
Compliance: Every periodic inspection
Effective Date: 30 September 1966
DCA/MS/117 Sliding Glass Hood - Inspection
Applicability: All model 880B and 885
Requirement: Morane Saulnier SB 35
Compliance: Next periodic inspection
Effective Date: 30 September 1966

DCA/MS/118 Nosewheel to Firewall - Inspection
Applicability: All model 880B and 885
Requirement: Morane Saulnier SB 38
Compliance: Every periodic inspection until modified
Effective Date: 30 September 1966

DCA/MS/119 Nosewheel to Firewall - Modification
Applicability: All model 880B and 885
Requirement: Morane Saulnier SB 38
Compliance: By 6 June 1975

DCA/MS/120 Fuselage - Carbon Monoxide Sealing - Inspection
Applicability: All model 880B and 885
Requirement: Morane Saulnier SB 44
Compliance: At each issue or renewal of the C of A
Effective Date: 30 September 1966

DCA/MS/121 Nose Landing Gear - Inspection
Applicability: All model 880B and 885
Requirement: Socata SB 48
Compliance: As detailed
Effective Date: 31 December 1969

DCA/MS/122 Slat Flexible Control with Lining - Replacement
Applicability: All model 880B and 885
Requirement: Socata SB 47
Compliance: Next overhaul
Effective Date: 31 December 1969

DCA/MS/124 Electrical Oil Pressure Assembly - Modification
Applicability: Model 880B S/N as listed in SB 49
Requirement: Socata SB 49
Compliance: By 1 March 1968
DCA/MS/125  Canopy Stops - Inspection
Applicability:  All model 880B and 885
Requirement:  Socata SB 75
Compliance:  Within the next 50 hours TIS and thereafter at intervals not exceeding 100 hours TIS
Effective Date:  31 May 1970

DCA/MS/126  Instrument Dial Glasses and Bezels - Inspection and Modification
Applicability:  All model 880B and 885
Requirement:  Socata SB 73
Compliance:  Within the next 50 hours TIS
Effective Date:  31 May 1970

DCA/MS/127  Battery Box - Modification
Applicability:  All model 880B
Requirement:  Socata SB 68
Compliance:  Within the next 100 hours TIS
Effective Date:  31 October 1970

DCA/MS/128  Fuel System - Inspection
Applicability:  Model 880B and 885 S/N as listed in SB 56
Requirement:  Socata SB 56 Parts 2 and 3
Compliance:  By 31 March 1971

DCA/MS/129  Grounding of Shielded Cables - Modification
Applicability:  Model 880B S/N as listed in SB 79
Requirement:  Socata SB 79
Compliance:  Within the next 50 hours TIS
Effective Date:  30 September 1971

DCA/MS/131  Horizontal Stabiliser Fittings - Modification
Applicability:  Model 880B and 885 S/N as listed in SB 12/1
Requirement:  Socata SB 12/1
Compliance:  By 1 June 1972

DCA/MS/132  Cowling Locks Adjustment - Inspection
Applicability:  All model 880B
Requirement:  Socata SB 107
Compliance:  Every periodic inspection
Effective Date:  30 April 1973
DCA/MS/133  Access Door in Upper Engine Cowling - Modification
Applicability:   Model 880B S/N as listed in SB 116
Requirement:    Socata SB 116
                (BV AI 75-1 also refers)
Compliance:     By 9 June 1975

DCA/MS/134  Carburettor Air Inlet Duct - Modification
Applicability:  All model 885 equipped with Avco-Lycoming or Franklin engines
Requirement:    Socata SB 125
Compliance:     Within the next 50 hours TIS
Effective Date: 30 September 1976

DCA/MS/135  Wing Attachment Fittings - Inspection
Applicability:  Model 880B and 885 prior to S/N 2963 and S/N 5050 through 5750
Requirement:    Inspect per Socata SB 129
                (BV AD 78-62(A) also refers)
Compliance:     Within the next 100 hours TIS
Effective Date: 26 May 1978

DCA/MS/136  Cancelled – DCA/MS/144 now refers

DCA/MS/137  Carburettor Air-inlet - Modification
Applicability:  Model MS 893E S/N 1 through 13072
Requirement:    Modify per Socata SB 137
                (BV AD 78-147(A) refers)
Compliance:     Within the next 50 hours TIS
Effective Date: 10 November 1978

DCA/MS/138  Airframe - Inspection
Applicability:  All model 880B, 885 and 893E
Requirement:    Inspect per Socata SB 133/2 column `A'. Repair and reprotect as prescribed before further flight
                (BV AD 82-89(A) refers)
Compliance:     Within the next 50 hours TIS or 4 months whichever is the sooner and thereafter at intervals not exceeding 100 hours TIS or 6 months whichever is the sooner
Effective Date: 10 September 1982
**DCA/MS/139  Horizontal Stabiliser - Inspection**

**Applicability:** All model MS 880B, 885, 893E and Rallye 235E

**Requirement:**
1. Inspect horizontal stabiliser spar and attachment fittings per Socata SB 145. Renew cracked parts before further flight.
2. Check attachment bolt installation and fit prescribed washers as necessary to achieve correct clamping.

**Compliance:**
1. Inspection - At intervals not exceeding 100 hours TIS
2. Check - Within the next 100 hours TIS and thereafter whenever fittings are replaced

**Effective Date:** 29 August 1985

**DCA/MS/140  Engine Mount - Inspection and Modification**

**Applicability:** All model 893E

**Requirement:** Remove identification plate, dye penetrant inspect engine mount and reposition plate per Socata SB 146. Repair cracks as prescribed before further flight

**Compliance:** Inspect and modify within next 100 hours TIS or by 31 December 1986, whichever is the sooner and thereafter inspect at intervals not exceeding 100 hours TIS

**Effective Date:** 29 August 1986

* DCA/MS/141A  Cancelled – EASA AD 2015-0203 refers

**Effective Date:** 21 October 2015

**DCA/MS/142  Nose Landing Gear Attachment - Inspection**

**Applicability** All Model 880, 885, 893 and 235E.

**Requirement:** To prevent failure of the nose landing gear, inspect per SOCATA Rallye SB 152 - 32 Rev1. Replace fittings found cracked per SB 152 - 32 Rev 1 before further flight. (BV AD 94-182 (A) R1 refers)

**Compliance:** Before further flight following a heavy landing or severe nose wheel shimmy.

**Effective Date:** 23 December 1994
DCA/MS/143  Rudder Control Clevis - Inspection

Applicability:  All Model 880, 885, 893 and 235E.

Requirement:  To prevent rudder control clevis failure, inspect the elevator and rudder governor control clevis per SOCATA Rallye SB 155-27. If the thickness at the bent section of the clevis is less than 0.045 in (1.1 mm), replace the clevis before further flight.  
(DGAC AD 2000-174(A) refers)

Compliance:  Within next 100 hours TIS.

Effective Date:  25 May 2000

DCA/MS/144  Engine Mount – Inspection and Modification

Applicability:  All model 110ST, 150, 180, MS 892, MS 893, and model MS 894A and MS 894E S/N 1005 through 2204, that are equipped with an engine mount ref. 887.51.0.030.0, 892.51.0.033.0, 85.51.0.001.0, 892.51.0.001.0, 892.51.0.053.0 or 892.51.0.060.0.  
This AD is not applicable to agricultural models, fitted with SOCATA kit n° OPT8098 9037.

Requirement:  To prevent failure of the engine mount, accomplish the following:-  
1. Inspect the engine mount per SOCATA Rallye SB 156-71. If a crack is found, the engine mount must be repaired or replaced before the next flight per SB 156-71. If the crack length is more than 6 mm the engine mount must be replaced.  
2. For the engine mounts with a riveted manufacturer’s plate, remove and bond the plate per SB 156-71.  
(DGAC AD 2001-400 refers)

Compliance:  1. Within next 50 hours TIS and thereafter at intervals not to exceed 50 hours TIS.  
2. Within next 50 hours TIS.

Effective Date:  25 October 2001

DCA/MS/145  Engine Mount - Replacement

Applicability:  Model A150, 150T, 150ST, 180T, MS 892 and MS 893 series equipped with engine mount ref. 892.51.0.035.0.

Requirement:  Replace the engine mount with an approved alternative. The engine mount ref. 892.51.0.035.0 may have been installed between 1 April 1978 and 6 November 1978, or by application of SOCATA Information Bulletin 39.  
(DGAC AD 1978-205(A) R1 refers)

Compliance:  Within next 5 hours

Effective Date:  25 October 2001
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* [2015-0203](#) NLG Axle – Inspection

**Effective Date:** 21 October 2015