Airworthiness Directive Schedule

Giders
Schleicher
28 October 2021

Notes:

1. This AD schedule is applicable to Schleicher gliders manufactured under the following EASA and Luftfahrt Bundesamt (LBA) Type Certificate (TC) Numbers held by Alexander Schleicher GmbH & Co.:

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2. The European Union Aviation Safety Agency (EASA) is the National Airworthiness Authority (NAA) responsible for the issue of State of Design Airworthiness Directives (ADs) for these gliders. State of Design ADs can be obtained directly from the EASA website at [http://ad.easa.europa.eu/](http://ad.easa.europa.eu/)

3. The date above indicates the amendment date of this schedule.

4. New or amended ADs are shown with an asterisk *

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The State of Design ADs listed below are available directly from the National Airworthiness Authority (NAA) websites. Links to NAA websites are available on the CAA website at https://www.aviation.govt.nz/aircraft/airworthiness/airworthiness-directives/links-to-state-of-design-

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If additional NZ ADs need to be issued when an unsafe condition is found to exist in an aircraft or aeronautical product in NZ, they will be added to the list below.

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DCA/SCH/101  Fuselage Members - Modification
Applicability: Model K7 S/N 1 through 935 and 984
Requirement: To preclude buckling of the fuselage steel tube members on either side of the rear fuselage, five tube replacements are necessary. These replacement tubes have a greater wall thickness and can be located by counting forward from stern post. Schleicher drawing Nr L-211-10-A4 dated 15 March 1961 and FAA AD 61-13-3 also cover their replacement.
1. Replace the following existing steel tube members using FAA Advisory Circular AC 43.13-1A procedures or manufacturer's recommendations.
2. Right side only third member, replace with 12 mm (15/32 in) outside diameter, 1 mm (.040 in) wall thickness tube.
3. Right and left side, fourth and sixth members, replace with 14 mm (9/16 in) outside diameter 1 mm (.040 in) wall thickness tubes.
Compliance: Before further flight.
Effective Date: 31 January 1967

DCA/SCH/102  Elevator Push Rod - Modification
Applicability: All model K7.
Requirement: Schleicher K7 Modification Nr 8.
Compliance: Next periodic inspection.
Effective Date: 31 January 1967

DCA/SCH/103  Freedom of Elevator Cable at Control Shaft - Modification
Applicability: All model Rhonlerche II.
Requirement: Rhonlerche II Modification Nr 9.
2. Part 2 - next periodic inspection.
Effective Date: 31 January 1967

DCA/SCH/104  Forked Air Brake Push Rod - Modification
Applicability: All model KA6 and K8.
Requirement: Schleicher KA6 Modification Nr 7 K8 Modification Nr 9.
Compliance: As detailed.
Effective Date: 31 January 1967

DCA/SCH/105  Automatic Elevator Connection - Inspection
Applicability: All model KA2, KA2B, KA6, K7 and K8.
Compliance: Every periodic inspection.
Effective Date: 31 January 1967
DCA/SCH/106  Push Pull Air Brake rod - Inspection
Applicability:  All model KA6 S/N up to 6067 except those which have complied with DCA/SCH/104.
Requirement:  LBA Airworthiness Information Nr 7/62.
Compliance:  Daily.
Effective Date:  31 January 1967

DCA/SCH/107  Control Stick Attachment Rear Cockpit - Inspection
Applicability:  All model KA2, KA2B, K7.
Requirement:  LBA Airworthiness Information Nr 2/63.
Compliance:  Daily.
Effective Date:  31 January 1967

DCA/SCH/108  Cancelled: Purpose fulfilled - Once only inspection

DCA/SCH/109  Elevator Horn Attachment - Inspection
Applicability:  All model K7 except those in which Auckland Gliding Club repair AGC/4/K7 has been embodied.
Requirement:  Cases have been reported of cracks being found between the elevator horn attachment blocks and the spar leading edge ply.
An inspection is to be made between the elevator horn attachment blocks and the spar leading edge ply; any cracks found are to be repaired by an approved scheme.
Compliance:  Every periodic inspection.
Effective Date:  31 January 1967

DCA/SCH/111  Wheel Brake Cable - Modification
Applicability:  All model ASK 13.
Requirement:  Schleicher ASK 13 modification Nr 2.
Compliance:  By 1 September 1968

DCA/SCH/112  Aileron Mass Balance - Modification
Applicability:  All model Rhonlerche II.
Requirement:  Schleicher Rhonlerche II modification Nr 12.
Compliance:  By 30 September 1968.
Effective:  1 September 1968
DCA/SCH/113 Larger Buffer Plates for the Rubber Buffer Landing Gear - Modification

Applicability: All model ASK 13.
Requirement: Schleicher ASK 13 modification Nr 3.
Compliance: By 30 April 1969

DCA/SCH/114 Suspension of the Thermo-Bottles below the Baggage Compartment - Modification

Applicability: All model KA6E S/N up to 4232 except 4226.
Requirement: Schleicher KA6E TN 17.
Compliance: By 30 April 1969

DCA/SCH/115A Aileron Control Safety Pin - Inspection

Applicability: All KA6 series.
Requirement: A case is reported in which the safety pin securing the aileron control rod to the aileron bellcrank fouled the edge of the fuselage aperture and movement of the control pulled the safety pin out with consequent risk of the control becoming disconnected. Subsequent investigation showed that the safety pin was non-standard and had probably been installed with the loop on the outside (towards the fuselage aperture edge).

1. All gliders of the subject type are to be inspected to ensure that the safety pin used to secure the aileron control rod to the aileron bellcrank is a standard manufacturer's part and that it is installed with the loop on the inside (away from the fuselage aperture edge).

2. On the surrounding surface of the aperture through which the control rods operate a placard is to be provided in red letters at least half an inch high to read "IMPORTANT - SAFETY PINS IN AILERON CONTROL MUST BE INSTALLED WITH LOOP ON INSIDE".

Compliance: Next periodic inspection.
Effective Date: 30 November 1971

DCA/SCH/116 Elevator Tube Spar - Modification

Applicability: All model KA6E.
Requirement: Schleicher KA6 TN 18.
Compliance: By 1 April 1971

DCA/SCH/117 Airbrake Bellcrank Inner Bearing - Modification

Applicability: Model K8 S/N 8119 and above.
Requirement: Schleicher K8 modification Nr 10.
Compliance: Within the next 25 hours TIS.
Effective Date: 30 November 1971
DCA/SCH/118  Push-Pull Rods Dust Cover - Modification
Applicability: All ASW 15 up to S/N 15139 inclusive.
Compliance: By 1 April 1972

DCA/SCH/119  Ventilation of Wing Nose Boxes - Modification
Applicability: Model ASW 15 S/N 15001 through 15163, except 15014, 15042, 15048 and 15162.
Compliance: By 1 April 1972

DCA/SCH/120  Cancelled: Purpose fulfilled - Once only inspection

DCA/SCH/121  Rudder Nose - Inspection
Applicability: Model ASW 15 S/N 15001 through 15183.
Compliance: Within the next 25 hours TIS.
Effective Date: 30 April 1973

DCA/SCH/122  Control Shaft, Strengthening - Inspection
Applicability: All model Rhonlerche II.
Requirement: Schleicher Rhonlerche II TN.
( LBA AD 75-166 refers)
Compliance: 1. Inspect before further flight.
2. Repair and modify in accordance with TN 13 Sheet 2 before further flight if cracks are found, but in any case modify not later than 1 July 1976.
Effective Date: 25 July 1975

DCA/SCH/123  Water Ballast System - Modification
Applicability: Model ASW 17 S/N 17001 through 17043.
( LBA AD 76-11 refers)
Compliance: Prior to next use of water ballast system.
Effective Date: 31 March 1976

DCA/SCH/124  Aileron Control Clevis Pin - Inspection
Applicability: All model Rhonlerche II.
( LBA AD 77-268 refers).
Compliance: Before further flight.
Effective Date: 12 October 1977
DCA/SCH/125  Inspection Panels - Modifications
Applicability: Model ASW 19 S/N 19001 through 19232.
Model ASW 20 S/N 20001 through 20113 except 20111.
Requirement: To prevent possible interference with aileron and flap controls:
Part 1 - tape each inspection panel to prevent its entry into fuselage.
Part 2 - embody modifications per Schleicher ASW 19 TN 7 or ASW 20 TN 4.
(LBA AD 78-303 refers)
Compliance: Part 1 - Prior to each flight until Part 2 accomplished.
Effective Date: 6 April 1979

DCA/SCH/126  Canopy Lock - Inspection
Applicability: All model K8B.
Requirement: Inspect canopy lock per Schleicher TN 21 and rework cam profile as necessary before further flight.
(LBA AD 80-158 refers)
Compliance: By 31 October 1980.
Effective Date: 26 September 1980

DCA/SCH/127  Aileron Hinge - Inspection
Applicability: All model ASW 19 and ASW 20.
Requirement: Inspect per LBA AD 81-74.
Compliance: Prior to next flight
Effective Date: 30 April 1981

DCA/SCH/128  Aileron and Flap Installations - Operating Limitation Placard and Inspection
Applicability: Model ASW 20 S/N 20001 through 20077 and 20086.
Requirement: 1. In clear view of pilot, affix placard which reads: "Vne LIMITED TO 108 KTS (200 KM/H)."
2. Inspect and rework as necessary per Schleicher ASW 20 TN 12.
(LBA AD 81-54 refers)
Compliance: 1. Placard - before further flight. May be removed after satisfactory completion of the inspection.
2. Inspection - by 30 September 1981.
Effective Date: 21 August 1981
DCA/SCH/129  Service Life - Inspection and Limitation
Applicability: All model ASW 15 and 15B.
Requirement: Accomplish inspection programme referred to in Schleicher ASW 15 TN 20. Any defects found must be rectified before further flight.
(LBA AD 81-91 refers)
Compliance: At 3000 hours TTIS and thereafter at intervals not exceeding 1000 hours TIS up to a maximum of 6000 hours TTIS.
Effective Date: 21 August 1981

DCA/SCH/130A  Elevator Actuator Bellcrank - Inspection
Applicability: All model ASW 15 and 15B.
Requirement: 1. Inspect per Schleicher ASW 15 TN 21. Renew cracked parts before further flight.
2. Modify per Schleicher ASW 15 TN 22.
(LBA AD 82-221 refers)
Compliance: 1. Inspection - within 100 flights or 100 hours TIS, whichever is the sooner, since inspection per DCA/SCH/130 and thereafter at intervals not exceeding 100 flights or 100 hours TIS whichever is the sooner, until modified.
Effective Date: DCA/SCH/130 - 26 February 1982
DCA/SCH/130A - 24 December 1982

DCA/SCH/131  Aileron/Airbrake Controls Installation - Inspection
Applicability: All model K7 and ASK 13.
Requirement: To preclude possibility of an in-flight malfunction, accomplish the following:
Gain access to aileron/airbrake operating lever shaft mounting brackets located on each inboard wing rib and visually inspect for cracks in area of lug bend radius adjacent to attachment bolts. Renew cracked parts before further flight.
Compliance: By 31 July 1982 and thereafter at intervals not exceeding six calendar months.
Effective Date: 9 July 1982

DCA/SCH/132A  Elevator Control System - Operating Limitation and Inspection
Applicability: All model ASW 19.
Requirement: To preclude possibility of elevator flutter, accomplish the following:
1. In clear view of pilot affix placard which reads - "Vne LIMITED TO 120 KTS (222 KM/H)". Placard may be removed when elevator profile modified per Schleicher TN 17.
2. Inspect fuselage bulkhead in front of control stick for security and freedom from damage. Rectify defects before further flight.
Compliance: 1. Placard - before further flight.
2. Inspection - at intervals not exceeding six calendar months and before further flight following a heavy landing.
Effective Date: DCA/SCH/132 - 17 February 1984
DCA/SCH/132A - 16 November 1984
DCA/SCH/133  Tow Release System - Inspection
Applicability: Model ASW 19 and 19B S/N 19001 through 19405; ASW 20 and 20L S/N 20001 through 20611; ASW 20B S/N 20620 through 20626.
Requirement: Inspect per Schleicher TN ASW 19/19B Nr 18 or ASW 20/20L/20B Nr 21 as appropriate. Incorrect installations to be rectified before further flight.
Compliance: By 31 October 1984.
Effective Date: 14 September 1984

DCA/SCH/134 Flight Controls and Safety Harness Installation - Inspection
Applicability: Models ASW 20 and 20L, S/N 20001 through 20611.
Requirement: To prevent unintentional pilot induced oscillations, accomplish the inspections and modifications prescribed in Schleicher TN ASW 20/20L Nr 30. Rectify defective installations before further flight.
Effective Date: 23 October 1987

DCA/SCH/135 Aileron Sealing - Inspection
Applicability: Models ASW 20, 20C, 20CL and 20L, S/N 20001 through 20860; also 20950 and 20951.
Requirement: To preclude possibility of aileron flutter, inspect per Schleicher TN ASW 20 Nr 31. Any defect or deficiency found, as described in "Action" paras 1.1 or 1.2, must be rectified before further flight.
Effective Date: 23 October 1987

DCA/SCH/136 Wing Spars - Inspection
Applicability: All model ASW 15 and 15B.
Requirement: To preclude possibility of in-flight wing failure, inspect per Schleicher ASW 15 TN 23, Actions 1.1 through 1.3 and 2.1 through 4. Inspection holes must be sealed and any damage repaired, as prescribed, before further flight.
Compliance: Actions 1.1 through 1.3 - Before further flight or by 30 June 1988, whichever is the sooner.
Actions 2.1 through 4 - By 31 December 1988.
Effective Date: 20 May 1988
DCA/SCH/137  Wing Spars - Inspection

Applicability:  All model ASW 17.

Requirement:  To preclude possibility of in-flight wing failure, inspect per Schleicher ASW 17 TN 12, Actions 1.1 through 1.3 and 2.1 through 4. Inspection holes must be sealed and any damage repaired, as prescribed before further flight.

(LBA AD 89-115 refers)

Compliance:  Actions 1.1 through 1.3 - By 30 November 1989.
Actions 2.1 through 4 - By 30 April 1990.

Effective Date:  1 November 1989

* DCA/SCH/138  Cancelled - EASA AD 2021-0230 refers

Effective Date:  28 October 2021

DCA/SCH/139  Airbrake Control - Inspection

Applicability:  All model ASK 13.

Requirement:  To prevent failure of the airbrake control bearing brackets inspect, repair and adjust as necessary per Schleicher ASK 13 TN 14.

(LBA AD 91-173 refers)

Compliance:  Within next 6 months and thereafter at intervals not exceeding 12 months.

Effective Date:  28 February 1992

DCA/SCH/140  Service Life - Inspection and Limitation

Applicability:  All Model ASW 17.

Requirement:  To extend service life to 12,000 hours implement the inspection program referred to in Schleicher ASW 17 TN 13. Any defects found must be rectified before further flight.

(LBA AD 93-124 refers)

Compliance:  By 3000 hours TTIS.

Effective Date:  24 December 1993
DCA/SCH/141  Canopy Retention and Flight Control Linkages - Inspection

**Applicability:** All model K8, K8B and K8C.

**Requirement:** To ensure the continued airworthiness of the glider accomplish the following parts of Schleicher K8 TN 24:-

- A1 Canopy retaining cord.
- A2 Rudder Pedals.
- A3 Elevator control Linkage.
- A4 Inspection of the fuselage tube frame and the control linkages for corrosion.

B1 Flight Manual Amendment.
B2 Diameter of the wing attachment pins.

If necessary rectify any deficiencies found per TN 24 before further flight.

(LBA AD 96-005 refers)

**Compliance:** Accomplish A1, A2, A3, and A4 by 30 June 1996 and thereafter at intervals not to exceed 12 months.

Accomplish B1 and B2 by 30 June 1996.

**Effective Date:** 15 March 1996

DCA/SCH/142  Service Life - Inspection and Limitation

**Applicability:** All model ASW 19.

**Requirement:** To extend service life to 12,000 hours accomplish the following:-

Implement inspection program per Schleicher ASW 19 TN 25, dated 21 October 1996. Any defects found must be rectified before further flight.

(LBA AD 97-010 refers)

**Compliance:** By 3000 hours TTIS until a maximum of 12,000 hours TTIS.

**Effective Date:** 6 June 1997

DCA/SCH/143  Service Life - Inspection and Limitation

**Applicability:** All model ASW 20 (all model variants).

**Requirement:** To extend service life to 12,000 hours accomplish the following:-

Amend the maintenance manual and implement the inspection program per Schleicher ASW 20 TN 39/2. Any defects found must be rectified before further flight.

(LBA AD 1998-255 refers)

**Compliance:** Amend maintenance manual by 31 December 1998. Initiate inspection program by 6000 hours TTIS until a maximum of 12,000 hours TTIS.

**Effective Date:** 6 June 1997
DCA/SCH/144  Service Life – Inspection and Limitation

Applicability: All model ASH 25 and ASH 25E.

Requirement: To extend service life to 12,000 hours accomplish the following:

1. Amend the aircraft flight manual and maintenance manual per Schleicher ASH 25 TN 14 or ASH 25E TN 12 as applicable, and implement the inspection program. Any defects found must be rectified before further flight.

2. Inspect per TN 14 or TN 12 as applicable, the elevator control linkage if a major repair to the landing gear or to the fin area has ever been accomplished. Rectify as necessary before further flight.

3. Incorporate additional safety device for the landing gear rear bolts per TN 14 or TN 12 as applicable.

(LBA AD 1998-486 refers)

Compliance:

1. Amend maintenance manual by 30 June 1999. Initiate inspection program by 6000 hours TTIS until a maximum of 12,000 hours TTIS.

2. By 12 April 1999.


Effective Date: 12 March 1999

DCA/SCH/145  Elevator Control Clearance - Inspection

Applicability: Model ASW 27 S/N 27002 through 27104.

Requirement: To prevent jamming of flight controls accomplish Schleicher ASW 27 TN 5.

(LBA AD 1999-283 refers)


Effective Date: 24 September 1999

DCA/SCH/146  Exhaust Muffler – Inspection

Applicability: Model ASH 25M.

Requirement: To prevent failure of the muffler, accomplish the following per Alexander Schleicher ASH 25 M Technical Note No. 15:

1. Determine which version of muffler is installed. Upgraded mufflers were marked with the letter “X”.

If a muffler without “X” marking is found installed and the operating time of the muffler is less than 40 hours, the CFRP-fairing of the muffler must be inspected for signs of overheating. If the muffler exceeds 40 hours engine operating time, the front side of the muffler behind the cover plate must also be inspected. If no discoloration is found, the engine may be operated for a further 2 hours and must be inspected every 2 hours up to a maximum of 60 hours engine operating time.

If damage or discoloration is found during any inspection, the muffler must be replaced before further flight.

If a muffler with an “X” marking is found installed, the muffler must be returned to Schleicher for an inspection at 100 hours total engine operating time or by 24 February 2001, whichever is the sooner.
2. Revise the Flight and Maintenance manual pages after the installation of the new muffler.

(LBA AD 1999-376 refers)

**Compliance:**
1. By 24 March 2000 and thereafter compliance is required at the times specified within the requirement of this airworthiness directive.
2. After the installation of the new muffler.

**Effective Date:** 24 February 2000

**DCA/SCH/147 Wing Ballast Tanks – Correction of CG Limits**

**Applicability:** Model ASW 27.

**Requirement:** The manufacturer has determined that the integral wing ballast tanks produce a larger nose-down moment than the soft ballast bags. To avoid a forward CG problem, accomplish the following:

1. Install the following placard and amend the flight manual per Schleicher ASW 27 Tech Note No 9.

```
WARNING: When water ballast is carried, pilots weighing 105kg or more (incl parachute) must use rearmost backrest hinge position
```

2. Determine empty CG position.

(LBA AD 2002-086 refers)

**Compliance:**
2. Within next 12 months.

**Effective Date:** 28 March 2002

**DCA/SCH/148 Fuel Line – Inspection**

**Applicability:** Model ASH 25M S/N up to an including 25233, excluding 25202, 25204, 25214, 25231, that are equipped with fuel injected engine IAE50R-AA.

**Requirement:** To prevent fuel leakage within the engine compartment and associated risk of fire, inspect the Fuel Line per Schleicher ASH 25 Mi TN No 22. If incorrect end fitting is found, replace before further flight.

(LBA AD 2003-129 refers)

**Compliance:** Before further flight.

**Effective Date:** 27 March 2003

**DCA/SCH/149 Rudder Pedal - Inspection**

**Applicability:** Centrair Model ASW 20F.

**Requirement:** To prevent possible failure of the rudder pedals and loss of rudder control inspect pedals per Centrair SB SN ASW 20F-23.

(LBA AD 2003-097 refers)

**Compliance:** By 30 November 2003.

**Effective Date:** 30 October 2003
DCA/SCH/150  Exhaust Insulation – Inspection

Applicability: Model ASH 26E aircraft, all S/N.

Requirement: To prevent an engine fire accomplish the following:

1. Inspect the oil sump air tube for damage, and the exhaust fairing heat insulation material and engine bay for oil contamination per the instructions in Schleicher Technical Note No. 6.

Replace defective parts and clean the engine bay as required, before further flight.

2. Install a placard in clear view near the engine oil tank with the following text:

<table>
<thead>
<tr>
<th>Use a funnel to fill the oil tank. Do not overfill the oil tank.</th>
</tr>
</thead>
<tbody>
<tr>
<td>Oil contamination of the engine compartment can result in an in-flight fire.</td>
</tr>
</tbody>
</table>

(LBA AD 98-347 refers)


Effective Date: 25 October 2007

DCA/SCH/151  Muffler – Inspection

Applicability: Model ASH 26E aircraft, all S/N.

Requirement: To prevent mufflers cracking, accomplish the following:

1. Inspect the muffler and establish if an upgraded muffler marked with a letter ‘X’ is fitted to the engine per Alexander Schleicher ASH 26 E Technical Note (TN) No. 8.

If the muffler is not marked with a letter ‘X’ accomplish requirement 2 of this AD.

If the muffler is marked with a letter ‘X’ accomplish requirement 3 of this AD.

2. For aircraft not fitted with an upgraded muffler marked with a letter ‘X’, inspect the CFRP fairing for overheating and the front side of the muffler behind the cover plate per TN No. 8.

If any damage or discoloration is found during any of these inspections, the muffler must be replaced before the next flight.

3. For aircraft fitted with an upgraded muffler marked with a letter ‘X’ accomplish the inspection requirement per TN No. 8.

(LBA AD 1998-311 refers)

Compliance: 1. Before further flight, unless already accomplished.

2. At 40 hours TTIS, or the next 2 hours TIS whichever is the later, and thereafter at intervals not to exceed 2 hours TIS until 60 hours TIS when the muffler shall be replaced per TN No. 8.

3. At 100 hours TIS or within the next 12 months whichever occurs sooner, unless already accomplished.

Effective Date: 25 October 2007
DCA/SCH/152  Flap Control Lever – Inspection

Applicability:  Model ASW 22, ASW 22 B, ASW 22 BL, ASH 25 and ASH 25 E aircraft, all S/N

Requirement:  To prevent an asymmetrical flap condition which could result in limited aircraft control and higher stall speeds, accomplish the following:

1. Inspect the flap control lever behind the rear cross tube per the instructions in Alexander Schleicher ASW 22 / B / BL Technical Note (TN) No. 16 or ASH 25 TN No.20 or ASH 25E TN No.28, as applicable.
   If any damage is found, replace the flap control lever with lever P/N 250.45.0070 per the instructions in the applicable Alexander Schleicher TN.

2. Replace the flap control lever located behind the rear cross tube with lever P/N 250.45.0070 per the instructions in the applicable Alexander Schleicher TN.

(EASA AD 2008-0059 refers)

2. At the next annual inspection or by 24 April 2009 whichever occurs sooner, unless already accomplished per requirement 1 of this AD.

Effective Date:  24 April 2008

DCA/SCH/153  Airworthiness Directive Compliance

Applicability:  Model ASK 21 aircraft, all S/N.

Note:  DCA/SCH/153 issued with the New Zealand type acceptance of the Schleicher ASK 21 glider under EASA Type Certificate A.221. The LBA ADs listed in this AD have no recurring requirements. Compliance with these LBA ADs are required before issue of a New Zealand Certificate of Airworthiness, or at the next ARA inspection after the effective date of this AD whichever is the sooner, unless previously accomplished.

Requirement:  Compliance with the following Luftfahrt-Bundesamt (LBA) Airworthiness Directives (as applicable) are required:

<table>
<thead>
<tr>
<th>LBA AD</th>
<th>Subject</th>
<th>Schleicher Technical Note (TN):</th>
</tr>
</thead>
<tbody>
<tr>
<td>1981-092</td>
<td>Rudder Pedals</td>
<td>(TN No. 5b refers)</td>
</tr>
<tr>
<td>1984-002</td>
<td>Tow Release Mechanism Cable</td>
<td>(TN No. 10 refers)</td>
</tr>
<tr>
<td>1984-032/2</td>
<td>Trim Ballast Weights</td>
<td>(TN No. 13a refers)</td>
</tr>
<tr>
<td>1984-180</td>
<td>Wheel Well</td>
<td>(TN No. 17 refers)</td>
</tr>
<tr>
<td>1986-236</td>
<td>Rudder Pedal Bracket Assembly</td>
<td>(TN No. 19 refers)</td>
</tr>
<tr>
<td>1988-002</td>
<td>Rudder Pedals, Airbrake Bellcrank and Rear Canopy Hinge</td>
<td>(TN No. 20 refers)</td>
</tr>
<tr>
<td>1990-350</td>
<td>Elevator Actuator Rod Parallel Rocker</td>
<td>(TN No. 8 and 22 refers)</td>
</tr>
<tr>
<td>1991-112</td>
<td>AFM Amendment – Spin Entry/Recovery</td>
<td>(TN No. 23 refers)</td>
</tr>
<tr>
<td>1993-186</td>
<td>Elevator Pushrod</td>
<td>(TN No. 26 refers)</td>
</tr>
<tr>
<td>1993-001/3</td>
<td>Ball and Socket Connectors with Lock Plates</td>
<td>(No reference service info.)</td>
</tr>
<tr>
<td>1994-001/2</td>
<td>Ball and Socket Connectors with Lock Cams</td>
<td>(No reference service info.)</td>
</tr>
<tr>
<td>1994-026</td>
<td>Maintenance Manual/Programme</td>
<td>(TN No. 24 refers)</td>
</tr>
</tbody>
</table>

Note:  Each part of this AD (each individual LBA AD) shall be certified in the aircraft log book separately.

Compliance:  Before issue of a New Zealand Certificate of Airworthiness, or at the next ARA inspection after the effective date of this AD whichever is the sooner, unless previously accomplished.

Effective Date:  26 November 2009
The State of Design ADs listed below are available directly from the National Airworthiness Authority (NAA) websites. Links to NAA websites are available on the CAA website at https://www.aviation.govt.nz/aircraft/airworthiness/airworthiness-directives/links-to-state-of-design-airworthiness-directives/
If additional NZ ADs need to be issued when an unsafe condition is found to exist in an aircraft or aeronautical product in NZ, they will be added to the list below.

2012-0246  Cancelled – EASA AD 2013-0091 refers
Effective Date:  26 April 2013

2013-0091  Automatic Elevator Control Connection – Inspection
Effective Date:  26 April 2013

2007-0042  RPM Sensor, Fuel Pump & Instruments – Improvements
Effective Date:  7 March 2007

2014-0264  Engine Mounts – Inspection
Effective Date:  23 December 2014

2013-0123  Spin Ballast Installation – AFM Amendment
Effective Date:  26 November 2015

2016-0192  Rudder Control System – AFM Amendment
Effective Date:  12 October 2016

2017-0136  Exhaust Silencer – Replacement
Applicability:  Schleicher ASK 21 Mi, ASW 22 BLE 50R, ASH 25 M (including those with sales designation ASH 25 Mi) and ASH 26 E powered gliders, all S/N.
Effective Date:  31 August 2017

2021-0187  Wing Root Ribs – Inspection
Applicability:  Schleicher ASW 15 gliders, S/N 15001 through to 15183.
Effective Date:  26 August 2021

* 2021-0230  Elevators – Inspection
Applicability:  Schleicher AS-K 13, ASK 16, ASK 16B, ASK 18, ASK 18 B, K 8, K 8 B, K 8 C and K 7 gliders and powered gliders, all S/N.
Schleicher Ka 6, Ka 6 B, Ka 6 BR, Ka 6 C, Ka 6 CR, Ka 6/0 gliders, all S/N.
Effective Date:  28 October 2021