Airworthiness Directive Schedule
Helicopters
Leonardo A109 Series
31 March 2022

Notes:
2. The European Union Aviation Safety Agency (EASA) is the National Airworthiness Authority (NAA) responsible for the issue of State of Design Airworthiness Directives (ADs) for these helicopters. State of Design ADs can be obtained directly from the EASA website at http://ad.easa.europa.eu/
3. The date above indicates the amendment date of this schedule.
4. New or amended ADs are shown with an asterisk *

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DCA/A109/1A  Airworthiness Directive Compliance at Initial Airworthiness Certificate Issue

**Applicability:** Agusta A109 series helicopters, all S/N.

**Note 1:** This AD revised to cancel ENAC AD 2006-001. The tail rotor blade inspections in Agusta BT 109-110 rev A have subsequently been introduced into chapter 5 of the A109A/A109All Maintenance Planning Manual (MPM).

**Requirement:** Compliance with the following Ente Nazionale Per L’Aviazione Civile (ENAC) Airworthiness Directives and European Aviation Safety Agency (EASA) Airworthiness Directives (as applicable) is required:

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Note 2: Each part of this AD (each individual ENAC and EASA AD) shall be certified in the aircraft log book separately.

Note 3: Manufacturer service information at later approved revisions is acceptable to comply with the requirements of this AD.

Compliance: Before issue of a New Zealand Certificate of Airworthiness, or at the next Review of Airworthiness after the effective date of this AD whichever is the sooner, unless previously accomplished.

Effective Date: DCA/A109/1 - 25 November 2010  
DCA/A109/1A - 27 September 2012

DCA/A109/2 Main Rotor Blades – Inspection

Applicability: Model A109E helicopters fitted with main rotor blades P/N 709-0103-01 (all dash numbers) with S/N all through 1428 with prefix A5 or EM.


Note: This AD supersedes ENAC AD 98-319 dated 15 September 1998.  
(ENAC AD 2000-573 refers)

Compliance: Within the next 10 hours TIS unless already accomplished and thereafter at intervals not to exceed 25 hours TIS.

Effective Date: 25 November 2010

DCA/A109/3 Engine Exhaust Ejectors – Inspection

Applicability: Model A109E helicopters, all S/N.

Requirement: To prevent engine exhaust ejector failure accomplish the following:

1. For model A109E helicopters, S/N all through 11036 accomplish the inspection/modification per part I in AGUSTA Bollettino Tecnico 109EP-3, dated 22 December 1998 or later EASA approved revisions.

2. For model A109E helicopters, S/N all through 11036 accomplish the inspection/modification per part II in AGUSTA Bollettino Tecnico 109EP-3, dated 22 December 1998 or later EASA approved revisions.

Note: This AD pertains to the inspection and modification of engine exhaust ejectors installation P/N 109-0601-51.  
(ENAC AD 1998-465 refers)


2. Within the next 10 hours TIS, unless already accomplished, and thereafter at intervals not to exceed 25 hours TIS for all A109E helicopters accomplish part III in AGUSTA Bollettino Tecnico 109EP-3.

Effective Date: 25 November 2010

DCA/A109/4 Cancelled – EASA AD 2020-0230 refers

Effective Date: 5 November 2020
**DCA/A109/5A  Hydraulic Pipes – Inspection**

**Applicability:** Model A109E aircraft, all S/N fitted with hydraulic pipes P/N 109-0761-64-103 or P/N 109-0761-65-103.

**Note 1:** This AD revised to reference Agusta Bollettino Tecnico (BT) 109EP-73 original issue or later EASA approved revisions.

**Requirement:** To prevent loss of hydraulic fluid from the number 1 hydraulic system due to the possibility of interference between the hydraulic pipes and the tail rotor control rod assembly, accomplish the following:

1. Inspect the hydraulic pipes with P/N 109-0761-64-103 and P/N 109-0761-65-103 per part I of Agusta Bollettino Tecnico (BT) 109EP-73 original issue or later EASA approved revisions. If interference is found between the hydraulic pipes and the tail rotor control rod assembly, accomplish the instructions in part II of Agusta BT 109EP-73, before further flight.


3. Hydraulic pipes P/N 109-0761-64-103 or P/N 109-0761-65-103 held as spares shall not be fitted to any aircraft.

**Note 2:** Accomplishment of requirement 2 is a terminating action to the repetitive inspection requirements of this AD. (EASA AD 2007-0231 refers)

**Compliance:**
1. Within the next 50 hours TIS unless previously accomplished and thereafter at intervals not to exceed 100 hours TIS.
2. By 21 May 2011 unless previously accomplished.
3. From 25 November 2010 (effective date of DCA/A109/5).

**Effective Date:**
- DCA/A109/5 - 25 November 2010
- DCA/A109/5A - 21 April 2011

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**DCA/A109/6  Cargo Hook Lever – Inspection**

**Applicability:**
- Model A109E aircraft fitted with a single hook installation P/N 109-0810-31-133 or double hook installation P/N 109-0810-31-133 or 109-0811-75-121 and with hook P/N 528-010-01.
- Model A109S aircraft fitted with a single hook installation P/N 109-0810-31-145, or double hook installation P/N 109-0810-31-145 or 109-0811-75-127 and with hook P/N 528-010-01.
- Model A109LUH aircraft fitted with a single hook installation P/N 109-0810-31-151 and with hook P/N 528-010-01.

**Requirement:**
To prevent failure of the cargo hook, inspect the lever P/N 232-028-00 for condition per the instructions in Agusta Alert SB No. 109EP-78, 109L-006 or 109S-12 as applicable, all at original issue or later EASA approved revisions. If the lever is cracked, repair as required before further hoist operations. (EASA AD 2007-0160-E refers)

**Note 1:** Compliance with the inspection requirement of this AD before every hoist mission may be accomplished by adding the inspection requirement to the tech log. The visual inspection may be subsequently accomplished under the provision in Part 43 Appendix A.1 (7) by the holder of a current pilot licence, if that person is rated on the aircraft, appropriately trained and authorised (Part 43, Subpart B refers), and the maintenance is recorded and certified as required by Part 43.

**Note 2:** Agusta S.p.A. is continuing the investigation to establish a terminating action.

**Compliance:** Before the next hoist operation or by 25 December 2010 whichever is the sooner, and thereafter before every hoist operation.

**Effective Date:** 25 November 2010
**DCA/A109/7 Exhaust Duct Clamps – Inspection**

**Applicability:** Model A109A, A109AII and A109C helicopters, all S/N.

**Requirement:** To prevent failure of the exhaust duct clamps, inspect the grooved clamps P/N 4606AC fixing the engine exhaust duct per the instructions in Agusta “Bollettino Tecnico” 109-123 dated 16 November 2006 or later EASA approved revisions. If any cracks and/or corrosions is found, discard and replace the affected grooved clamp.

(EASA AD 2007-0041 refers)

**Compliance:** Within the next 50 hours TIS or by 25 January 2011 whichever occurs sooner and thereafter at intervals not to exceed 300 hours TIS, or every 12 months whichever occurs sooner.

**Effective Date:** 25 November 2010

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**DCA/A109/8 Tail Rotor Blades – Inspection**

**Applicability:** Model Agusta A109E helicopters, all S/N fitted with Tail Rotor Blades P/N 109-8132-01-111.

**Requirement:** To prevent failure of the tail rotor blades accomplish the inspections and corrective actions as required, at the thresholds and intervals specified in Agusta BT 109EP-30 revision C or later EASA approved revisions.

**Note:** Agusta Bollettino Tecnico 109EP-30 revision C and the Agusta A109E Maintenance Planning revision 1 or later approved revisions pertains to the subject of this AD.

(EASA AD 2007-0010 refers)

**Compliance:** At the thresholds and intervals specified in Agusta BT 109EP-30 unless previously accomplished.

**Effective Date:** 25 November 2010

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**DCA/A109/9 Cancelled – AD limited to A109E, S/N 11671 only**

**Effective Date:** 25 August 2011

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**DCA/A109/10 Cancelled – EASA AD 2017-0085-E refers**

**Effective Date:** 16 May 2017

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**DCA/A109/11 MLG Actuator Bracket Attachment Bolts – Inspection**

**Applicability:** Model A109A, A109AII and A109C aircraft, all S/N.

**Requirement:** To prevent failure of the MLG, accomplish the following:

1. Replace the RH and LH MLG actuator bracket attachment bolts P/N NAS624H8 and P/N NAS624H10 and torque per the instructions in part I of AgustaWestland BT 109-133, dated 04 November 2011 or later approved revisions.

2. Inspect the RH and LH MLG actuator bracket attachment bolts P/N NAS624H8 and P/N NAS624H10 per the instructions in part II of BT 109-133.

If any defects are found with any of the MLG actuator bracket attachment bolts, replace all the bolts on the affected side (RH or LH MLG, as applicable) before further flight per the instructions in part I of BT 109-133.

(EASA AD 2011-0236 correction refers)

**Compliance:**

1. Within the next 100 hours TIS or by 26 June 2012 whichever occurs sooner.

2. Within the next 100 hours TIS after replacement of the MLG actuator bracket attachment bolts per requirement 1 of this AD, and thereafter at intervals not to exceed 100 hours TIS or 12 months whichever occurs sooner.

**Note:** The replacement of MLG actuator bracket attachment bolts per the requirement of this AD is not a terminating action for the repetitive inspections mandated by this AD.

**Effective Date:** 26 January 2012
DCA/A109/12  Tail Rotor Duplex Bearing Ring Nut – Inspection

Applicability:  Model A109E helicopters, S/N 11002 through to 11807, except 11796
Model A109LUH helicopters, all S/N
Model A109S, all S/N
Model AW109SP, S/N 22202 through to 22278, except 22239, 22264, 22266, 22272, 22273, 22275 and 22277
Model A109C helicopters, all S/N
Model A109K2 helicopters, all S/N

Requirement:  To prevent loss of the tail rotor duplex bearing ring nut, accomplish the requirements in EASA AD 2012-0195-E.


(EASA AD 2012-0195-E refers)

Compliance:  At the compliance times specified in EASA AD 2012-0195-E.

Effective Date:  27 September 2012
The State of Design ADs listed below are available directly from the National Airworthiness Authority (NAA) websites. Links to NAA websites are available on the CAA website at https://www.aviation.govt.nz/aircraft/airworthiness/airworthiness-directives/links-to-state-of-design-airworthiness-directives/

If additional NZ ADs need to be issued when an unsafe condition is found to exist in an aircraft or aeronautical product in NZ, they will be added to the list below.

2013-0208  Cancelled – EASA AD 2013-0290 refers
            Effective Date: 23 December 2013

2013-0009  Blade Retaining Bolts – Inspection
            Effective Date: 25 January 2013

2013-0118  Cancelled – EASA AD 2015-0096 refers
            Effective Date: 12 June 2015

2013-0225-E Tail Rotor Driveshaft – Inspection
                        A109E helicopters, all S/N up to 11832 inclusive, except S/N 11796, from 11808 to
                        11810 inclusive and from S/N 11812 to 11829 inclusive.
                        A109K2 helicopters, all S/N.
                        A109LUH helicopters, all S/N.
                        A109S helicopters, all S/N.
                        AW109SP helicopters, all S/N up to 22316 inclusive, except S/N 22284, 22286,
                        22307 and 22308.
                        A119 and AW119MKII helicopters, all S/N up to 14811 inclusive, except 14805 and
                        14807.
            Effective Date: 21 September 2013

2013-0265-E Main Rotor Swashplate Support Nut – Inspection
                        A109E helicopters, all S/N.
                        A109K2 helicopters, all S/N.
                        A109LUH helicopters, all S/N.
                        A109S helicopters, all S/N.
                        AW109SP helicopters, all S/N.
                        A119 and AW119MKII helicopters, all S/N.
            Effective Date: 1 November 2013
2013-0290  Main Rotor Lag Damper – Inspection
Effective Date: 23 December 2013

2014-0037  Main Rotor Driveshaft Nuts – Inspection
A109K2 helicopters, all S/N.
A109LUH helicopters, all S/N.
A109S helicopters, all S/N.
AW109SP helicopters, all S/N up to 22327 inclusive, except S/N 22284, 22286, 22307, 22321, 22323 and 22326.
Effective Date: 28 February 2014

2014-0150  Cancelled – EASA AD 2019-0294 refers
Effective Date: 18 December 2019

2014-0238-E  MGB Support Assembly – Inspection
Effective Date: 4 November 2014

2015-0025-E  Main Rotor Blades – Inspection
Applicability: A109A helicopters, S/N 7154 through to 7255 inclusive, and
A109AII helicopters, all S/N.
Effective Date: 20 February 2015

2015-0035-E  Tail Rotor Pitch Control Link Assembly – Inspection
Effective Date: 3 March 2015

2015-0054  Drive Shaft Assembly – Inspection
Effective Date: 10 April 2015
2015-0096 Main Gearbox Gleason Crown – Inspection  
   **Applicability:** A109E, A109K2, A119 and AW119MKII helicopters, all S/N.  
   **Effective Date:** 12 June 2015

2015-0097 Cancelled – EASA AD 2020-0142 refers  
   **Effective Date:** 30 July 2020

2015-0190R1 Cancelled – EASA AD 2016-0213 refers  
   **Effective Date:** 31 October 2016

2010-0190-E Passenger Sliding Door Locks – Inspection  
   **Applicability:** AW109SP helicopters, S/N 22202, and S/N 22204 through to 22211.  
   **Effective Date:** 22 October 2015

2015-0022 Airworthiness Limitations – Amendment  
   **Applicability:** AW109SP helicopters, S/N 22202, and S/N 22204 through to 22213.  
   **Effective Date:** 22 October 2015

2015-0227 Passenger Cabin Sliding Doors – AFM Amendment  
   **Applicability:** A109S helicopters, all S/N.  
   **Effective Date:** 3 December 2015

2016-0173-E Tail Rotor Blade Retention Bolts – Inspection  
   **Applicability:** A109E, A109K2, A109LUH, A109S, A119, AW109SP and AW119MKII helicopters, all S/N.  
   **Effective Date:** 26 August 2016

2016-0213 Main Rotor Blades – Inspection  
   **Applicability:** A109A and A109AII helicopters, all S/N.  
   **Effective Date:** 31 October 2016

2016-0261R1 Fire Extinguisher Bottles – Inspection  
   **Applicability:** A109LUH, A109E, A109S and AW109SP helicopters, all S/N.  
   **Note:** EASA AD 2016-0261R1 revised to expand the AD applicability to include additional affected fire extinguisher kit part numbers. The AD is still considered an interim action and further AD action may follow.  
   **Effective Date:** EASA AD 2016-0261 - 4 January 2017  
   EASA AD 2016-0261R1 - 27 February 2020
2017-0025  Hoist – Inspection
Applicability: AW109SP helicopters, all S/N.
Effective Date: 21 February 2017

2017-0046-E  Engine and Transmission Oil Cooling System – Inspection
Effective Date: 14 March 2017

2017-0085-E  Elevator Assembly – Inspection
Applicability: A109S and AW109SP helicopters, all S/N.
Effective Date: 16 May 2017

2017-0176-E  Main Rotor Blades – Inspection
Effective Date: 18 September 2017

2018-0053-E  Swashplate Support - Inspection
Applicability: AW109SP helicopters, all S/N.
Effective Date: 22 March 2018

2018-0120-E  Cancelled - EASA AD 2018-0149-E
Effective Date: 17 July 2018

2018-0149-E  Electrical Cables - Inspection
AW109SP helicopters, all S/N up to 22386 inclusive, except S/N 22375 and 22376.
Effective Date: 17 July 2018

2018-0205  Main Rotor Floating Ring Assembly - Inspection
Effective Date: 21 September 2018

2018-0280  Mixing Control Connecting Link – Inspection
AW109SP helicopters, all S/N.
Effective Date: 24 December 2018

2019-0182  Enhanced MLG Strut Assembly – Inspection
Effective Date: 29 August 2019
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<td>2020-0142</td>
<td>Tail Rotor Drive Slider Assembly Pitch Control – Inspection</td>
<td>A109A and A109All helicopters, all S/N.</td>
<td>30 July 2020</td>
</tr>
<tr>
<td>2020-0256</td>
<td>Centre Fuselage Frame Assembly – Inspection</td>
<td>A109E helicopters, S/N 11001 through to 11674.</td>
<td>1 December 2020</td>
</tr>
<tr>
<td>2021-0031</td>
<td>Gearbox Assembly – Inspection</td>
<td>A109E helicopters, all S/N up to and including 11160.</td>
<td>5 February 2021</td>
</tr>
<tr>
<td>2021-0065</td>
<td>Wiring Protection – Modification</td>
<td>AW109SP helicopters, S/N 22375 and 22387 through to 22422, except S/N 22404, 22405, 22407, 22409, 22413, 22415, 22419, 22420 and 22421. A109S helicopters, S/N 22707 through to 22717, 22721, 22723, 22729 and 22731 through to 22734.</td>
<td>25 March 2021</td>
</tr>
</tbody>
</table>
2021-0144  Tail Rotor Shaft Assembly – Inspection  
Effective Date:  1 July 2021  

2021-0179  Hoist Support Assembly – Inspection  
Applicability:  AW109SP helicopters, all S/N.  
Effective Date:  10 August 2021  

2021-0255(Correction)  Control Rods and Levers – Inspection  
Applicability:  A109S helicopters fitted with TREKKER KIT, S/N 22735, 22736 and 22737; and  
AW109SP helicopters, S/N 22407, 22408, 22409, 22412, 22414 to 22427 included,  
and 22429.  
Effective Date:  29 November 2021  

* 2022-0037(Correction)  Main Rotor Rotating Scissor Assembly – Inspection  
Effective Date:  EASA AD 2022-0037 - 21 March 2022  
              EASA AD 2022-0037 (Correction) - 31 March 2022