

Airworthiness Directive Schedule

Helicopters

Sikorsky Aircraft S-55 Series

25 August 2005

The date above indicates the amendment date of this schedule.

New or amended ADs are shown with an asterisk *

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*** DCA/S-55/1 Airworthiness Directive Compliance at Initial Airworthiness Certificate Issue****Applicability:** Model S-55 aircraft. All S/Ns.**Requirement:** Compliance with the following FAA Airworthiness Directives (as applicable) is required.

FAA Airworthiness Directives:

- 75-22-12 - In-flight Power Loss
- 75-04-02 - Main Rotor Spindles
- 73-16-01 - Engine Turbine Failure
- 71-26-03 - Servo Support Bracket
- 70-12-04 - Main Rotor Shaft Cracks
- 67-29-07 - Azimuth Control Rod Assembly
- 67-17-06 - Pylon Lower Ring
- 66-04-03 - Main Rotor Blades
- 65-08-02 - Main Rotor Blades
- 63-05-02 - Main Rotor Blades
- 60-13-04 - Rotor Shank Rod
- 56-23-03 - Horizontal Hinge Pin Assembly
- 56-16-03 - Main Rotor Blade Inspection
- 55-25-04 - Tail Cone Skin Reinforcements
- 54-20-02 - Support Assembly Cracks
- 54-19-02 - Servo Safety Clip
- 54-16-01 - Main Transmission
- 54-13-01 - Star Assembly Bolt Replacement
- 54-01-03 - Fuel Strainer Screen

Note: Each part of this AD (each individual FAA AD) shall be certified in the aircraft log book separately.**Compliance:** Before issue of New Zealand Certificate of Airworthiness and thereafter at each Annual Review of Airworthiness. Repetitive inspections to be accomplished at intervals not exceeding the times specified in the FAA Airworthiness Directives.**Effective Date:** 25 August 2005